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Checklist für Diamond DA 42 Twin Star

Edition #: 12 Edition date: 01.07.2006

Please observe:

In order to simplify the production process the following method has been chosen:

Normal Checklist, Emergency Checklist and Abnormal Checklist are produced separately, and therefore each will have its own "edition #"; "edition date" and pagination sequence.

The file you are receiving hereby, however, combines all three sections: Normal Checklist, Emergency Checklist and Abnormal Checklist.

For each of the 3 sections **all** pages of a new edition will have the same new "edition #" and "edition date", even if only one page was amended and all other pages still have the same, unchanged content.

Therefore the "List of Effective Pages" (LEP) is provided. It is here where you can see whether a particular page was amended. Pages which have been amended by a new edition will be marked yellow. For all other pages you will see which original "edition #" (and of course any higher "edition #") is still valid.

It is now up to you, whether you chose to print out the complete new checklist (thereby making papermills happy) or to print out only those pages which have been amended (thereby stressing brain cells how to do this).

Have a lot of nice flights and happy landings!

Peter Schmidleitner

Comments explaining Edition # 12 are on page 2 of this document

Checklist DA42 Twin Star - LEP

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Comments explaining Edition # 11.5

Normal Checklist

Some PREFLIGHT EXTERIOR items rearranged

FMS setup: item "Radios" expanded

Autopilot test and Altimeter setting added to "check after engine start"

Check of electrical trim, ice protection and parking brake release added to "before TKOF check"

"Start key REMOVED" added to Parking Check

Page 9: stall speed increment for ice added

Emergency Checklist

Oxygen system related items added.

Abnormal Checklist

DEIC PRES LO procedure amended according "fiki" supplement.

Emergency Checklist Abnormal Checklist

Layout of these checklists has been completely revised!

As you might know there is an inconsistency in the DA42 AFM since it is not arranged according to the principle

"red light - red chapter", "yellow light - yellow chapter".

This originally caused me to develop that somewhat awkward introductory table (with that red and yellow blocks) on page 2 of the previous emergency checklist.

Despite of this AFM structure I decided now to revise the DA42 checklist, and this checklist now adheres to the principle "red light – red checklist", "yellow light – yellow checklist".

In addition (since the cover sheet of the Emergency Checklist has the heading "Emergency + Abnormal Checklist") I changed the pagination and pages are now numbered without starting the Abnormal Checklist with another "1".

Comments explaining Edition # 12

This new edition is the result of a major exercise to streamline the checklists for all the 5 Diamond aircraft types (DA40-180, DA40 TDI, DA40-180 G1000, DA40 TDI G1000 and DA42) and to bring them to a common layout and standard.

In addition:

Emergency Checklist

All references to "12.500 ft" in the "Oxygen System" checklists have been changes to "10.000 ft" in order to comply with a revision of the relevant AFM supplement S04.

Note:

In the future the system of assigning "Edition #" will be as follows:

- if the revision affects all types, a new edition # (without a decimal figure) will be assigned to all of the checklists
- if the revision does not affect all types, the affected checklists will get subsequent "decimal figures" until a major revision affecting all checklists is issued.

NORMAL CHECKLIST



This checklist is compiled according the guidelines of GAMA Specification No.1, SECTION 3, para 3.5, SECTION 3A, para 3A.5 and SECTION 4, para 4.5.

The "Amplified Normal Procedures", "Amplified Emergency Procedures" and "Amplified Abnormal Procedures" according GAMA Specification No. 1 are in the DA42 Airplane Flight Manual Chapters 4A, 3 and 4B.

This checklist is not a substitute for and does not supersede any portion of the current approved Airplane Flight Manual or any of its supplements, or any training or procedures required by any regulatory or advisory bodies. Use of this checklist is at the user's sole risk and discretion. Diamond Aircraft shall have no liability for any damages, injury or death resulting from its use.

All such terms and conditions shall be deemed to be accepted in full by continued use of this checklist.

If you do not understand, or if you disagree with, any of the above terms and conditions, further use of this checklist is not permitted. Further use of this checklist is prohibited in any jurisdiction that does not give effect to all provisions of these terms and conditions.

Use of the electronic checklist (if available):

Before using the electronic checklist on the G1000 the following sections have to be completed using this paper checklist:

- Preflight interior + exterior
- Preflight exterior
- Check before engine start items 1 to 21 (may be completed by heart).

This checklist also serves as a back up for the electronic checklist in case the G1000 MFD is not available.

DA42 Twin Star

Attention!

For refuelling with JET A1 no additives (e.g. "Aerojet") are permitted.

- if optional ice protection is installed
- if optional AUX tanks are installed

PREFLIGHT INTERIOR + EXTERIOR.

- 1 Check airplane documents
- 2 Remove pitot cover
- Check interior for foreign objects
- Check circuit breakers
- Start kev PULLED OUT
- Gear selector CHECKED DOWN
- Electric Master ON
- Gear 3 greens CHECKED
- Check fuel quantity + temp
- 10 ** Fuel transfer ON if L/R AUX FUEL E caution ON: AUX tank(s) empty Fuel transfer OFF
- 11 * Check de-ice fluid quantity
- 12 * De-ice HIGH (if required)
- 13 External (* ice) lights ON
- 14 Pitot heat ON
- 15 Check external lights
- 16 Check stall warning
- 17 Check pitot/static tube heat
- 18 * Check de-ice function
- 19 Pitot heat OFF
- 20 External lights OFF
- 21 * De-ice, ice lights OFF
- 22 Electric Master OFF

PREFLIGHT EXTERIOR

Canopy left side

Left main gear

Strut (min 4cm bare piston) & downlock

Tire condition, pressure (4,5 bar), position mark

Brake, hydraulic line

Gear door & linkage

Left engine nacelle

Drain cascolator

3 air inlets / 2 air outlets

Spinner, propeller

Gearbox oil level

Engine oil level

Cowling

Nacelle underside

Venting pipe

Fxhaust

** Check AUX tank full?

Left wing

Wing leading edge, top- and bottom surface

Tank drain

Stall warning

Tank air vent

Fuel filler cap

Pitot, static probe (cover

removed)

Wing tip, position light

Static dischargers

Aileron (freedom of movement,

hinges, control linkage,

security) Wing flap

Fuel cooler air in- & outlet

** AUX tank vent

** Drain AUX tank

Left fuselage

Step

Rear cabin door

Fuselage left side

Antennas

Tail

Elevator & rudder (freedom of movement, hinges) Elevator & rudder trim - tabs Tail skid & lower fin Static dischargers

Right fuselage

Fuselage right side Rear window Step

Right wing

Fuel cooler air in- & outlet

** AUX tank vent

** Drain AUX tank

Wing flap

Aileron (freedom of movement,
hinges, control linkage,
security)

Static dischargers

Wing tip, position light

Wing leading edge, top- and
bottom surface

Fuel filler cap

Tank air vent

Canopy right side

Tank drain

Right engine nacelle

** Check AUX tank full ?
3 air inlets / 2 air outlets
Spinner, propeller
Gearbox oil level
Engine oil level
Cowling
Nacelle underside
Venting pipe
Exhaust
Drain cascolator

Ventilation air inlet

Right main gear

Strut (min 4cm bare piston) & downlock
Tire condition, pressure (4,5 bar), position mark
Brake, hydraulic line
Gear door & linkage

Nose section

* De-ice fluid tank
L + R front baggage door locked
OAT sensor
EPU connection
Landing / Taxi light

Nose gear

Strut (min 15cm bare piston) & lock
Tire condition, pressure (6 bar), position mark
Gear door & linkage

Chocks removed
Tow bar removed

CHECK BEFORE ENGINE START

1	Preflight checkCOMPLETED	1
2	Baggage and tow bar SECURED	2
3	Fuel selectors (2)ON	3
4	Power levers (2) IDLE	4
5	Parking brakeSET	5
6	Alternate Air	6
7	Manual gear extension handle PUSHED	7
8	Gear selector DOWN	8
9	Avionic master OFF	9
10	Electric master OFF	10
11	Engine masters (2) OFF	11
12	Pitot heat OFF	12
13	Alternate static	13
14	Alternators (2) ON	14
15	ECU swap (2)AUTO	15
16	All light switches OFF	16
17	Emergency switchOFF/GUARDED	17
18	ELTARMED	18
19	Circuit breakersCHECKED IN	19
20	Flap selectorUP	20
	If starting with sytample powers	

If starting with external power:

	a Prop area CHECK CLEAR a	
	b External powerCONNECT b	
21	Electric master ON	21
	Rudder pedals ADJUSTED	22
	Flight controls CHECKED	23
	Trims CHECKED	
	Gear warning, fire detectorTEST	25
26	* De-ice ANNUN TEST ON	26
27	* DEICE LVL LO caution. CHECKED ON if applic.	27
28	* Windshield de-icing PUMP 1 + 2 CHECKED	28

Checklist continued next page

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CHECK BEFORE ENGINE START continued

29	FlapsLDG	29
30	Variable elevator backstop CHECK	30
	Control stick	
31	FlapsUP	31
32	Passengers INSTRUCTED	32
33	Seat belts FASTENED	33
34	Rear door CLOSED and LATCHED	34
35	Front CanopyPOS 1 or 2	35
36	G1000POWERED, ACKNOWLEDGED	36
37	PFD/MFD BACKUP MODE	37
38	MFD ENGINE - FUEL	38
39	Fuel Quantity CHECKED, RESET/SET if requ.	39
40	Fuel temperature CHECKED	40
41	Total time in serviceNOTED	41
42	MFDENGINE - SYSTEM	42
43	* DEIC PRESS LO caution CHECKED ON	43
44	* De-ice ANNUN TEST OFF	44
45	Power levers (2) IDLE	45
46	ACL (strobe) ON	46

End of Checklist

ENGINE START PROCEDURE

Normal sequence: first start LH engine

Engine Master	ON
Annunciations / Eng.Instr	CHECKED
Glow indication	OFF
Propeller area	CLEAR
Start key	START
Oil pressure OUTSIDE RED	within 3 sec
Voltage, Electrical load CHECK 1	INDICATION
Annunciations / Eng.Instr	CHECK

If external power was used:

External power......DISCONNECT

Start RH engine, procedure as above

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CHECK AFTER ENGINE START

	1	Oil pressure CHECKED	1
	2	RPM 900 +/- 20 CHECKED	2
	3	Warm up time START	3
		Warm up: Idle2 minutes 1400RPMuntil Oil > 50℃ and Coolant > 60℃	
	4	Fuel selectors (2)X-FEED	4
	5	Pitot heatON, annunciation + Amps checked	5
	6	Pitot heat OFF	6
	7	PFD/MFD NORMAL MODE	7
	8	Avionics master ON	8
-		FMS SETUP	
		 I nitialize profile (AUX 4, MAP, MFD FPL, PFD FPL) F light plan R adios (COM, NAV, ADF, DME, CDI, BRG 1/2) P erformance (speed bugs) 	
	9	FMS setupCOMPLETED	9

AUTOPILOT TEST

DISCONN press, check electric trim not working AP ON, check overpowering servos DISCONN press, check AP off

	DISCONN press, eneck Ar on	
10	Autopilot testCOMPLETED	10
11	Flood light CHECKED, ON as required	11
	Position lightsON as required	
13	Fuel Selectors (2) ON	13
	Altimeters (3) SET	
15	Standby horizon CHECKED	15
	TransponderCODE / MODE CHECKED	
17	Parking brakeRELEASED	17

End of Checklist

DURING TAXI

Check Brakes Check nose wheel steering Check flight instruments

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BEFORE TAKE OFF CHECK

1	Parking brakeSET	1
2	Seat belts FASTENED	2
3	Rear door CLOSED + LATCHED	3
4	Front canopy	4
5	Front baggage doors CHECKED CLOSED	5
6	Door warning light OFF	6
7	Engine instruments CHECKED	7
8	Fuel temperature (Diesel min. +5°) CHECKED	8
9	Circuit breakers CHECKED	9
10	Electric elevator trim CHECKED, T/O SET	10
11	Fuel selectors (2)CHECKED ON	11
12	Rudder trim AS REQUIRED	12
13	Flaps CHECKED UP	13
14	Flight controls CHECKED	14
15	Power levers (2) IDLE	15
16	ECU test (2) PERFORM	16

ECU TEST

ECU test button	press and hold
"L/R ECU A/B fail"	ON / RPM increasing / OFF
"L/R ECU B fail"	ON / prop cycling / OFF
"L/R ECU A fail"	ON / prop cycling / OFF
<i>RPM</i>	decrease to idle
FCU test button	release

17	ECU swap (2) ECU B, ENGINES CHECKED	17
18	ECU swap (2)AUTO	18
19	Pitot heat AS REQUIRED	19
20	* Ice protection AS REQUIRED	20
21	TransponderCODE / MODE CHECKED	21
22	Parking brakeRELEASED	22

End of Checklist

LINE UP PROCEDURE

Landing light	ON
Approach sector	
Runway	IDENTIFIED
Power lever max (100% / 10 sec)	
CHECK LOAD / RPM	/ FUEL FLOW / OP

AFTER TAKE-OFF PROCEDURE

Brakes	APPLY
Gear	UP
Landing light	OFF

CLIMB TO CRUISE CHECK

1	Gear CHECKED UP	1
2	Flaps CHECKED UP	2
3	Landing light CHECKED OFF	3
	End of Checklist	

PERIODICALLY DURING CRUISE

Fuel Radio Engine Direction Altitude

DESCENT / APPROACH CHECK

	1	Landing data RECEIVED	1
	2	Altimeters (3) SET	2
		COM / NAV / FMS SET	3
		SeatbeltsFASTENED	4
	5	Fuel selectors (2)CHECKED ON	5
	6	Parking brakeCHECKED RELEASED	6
	7	Gear warning horn CHECKED	7
,		End of Checklist	

BEFORE LANDING PROCEDURE

Downwind, latest base leg:	
Flaps	APP
GearDOWN, (CHECK 3 GREENS
Landing light	ON

On final when landing assured:

FINAL CHECK

1	FlapsLDG	1
2	Gear 3 GREENS CHECKED	2

GO AROUND PROCEDURE

Power	MAX
Flaps	
Positive rate of climb:	
GearGear	UP
Continue with take-off profile	
At safe altitude:	
Flaps	UP
Landing light	OFF

AFTER LANDING CHECK

When clear of runway

1	FlapsUP	1
2	Pitot heat OFF	2
	Alternate air	
	* De-ice systems OFF	
	Landing/Taxi lightAS REQUIRED	

End of Checklist

PARKING CHECK

1	Parking brakeSET	1
2	Power levers (2) IDLE for 2 min.	2
3	ELT 121,5 CHECKED	3
4	Engine / System page CHECKED	4
5	Engine / Fuel page TTL TIME IN SVC NOTED	5
6	Avionic master OFF	6
7	Electrical consumers except ACL (strobe) OFF	7
8	Engine Masters (2) OFF	8
9	ACL (strobe) OFF	9
10	Electric Master OFF	10
11	Interior lightCHECKED OFF	11
12	Start keyREMOVED	12

End of Checklist

SECURING THE AIRCRAFT

Release parking brake, use chocks. Attach tie down ropes to mooring points.

OPERATING SPEEDS KIAS for MTOM 1785

	1400 kg	1785 kg	
Stalling speed (V _{S0}) Flaps LDG	49	57	
Stalling speed (V _S) Flaps APP	53	61	
Stalling speed (V _S) clean	56	64	
In Ice: + 4 K	t		
Best gliding angle (Flaps UP)	8	2	
Best angle of climb (V _x)	7	9	
Best rate of climb (V _Y)	7	9	
Best rate of climb 1-eng. (V _{YSE})	8	2	
Min. control speed (V _{MCA})	6	8	
Min. control speed for $TRG(V_{SSE})$	8	2	
Min. control speed (V_{MCA}) in ice	7	2	
Operating speed in ice	121 -	- 160	
Cruising climb speed	8	6	
Rotation speed	72		
Max. flap speed (V _{FE}) Flaps APP	137		
Max. flap speed (V _{FE}) Flaps LDG	111		
Max. LG extension (V _{LOE})	194		
Max. LG extended (V _{LE})	194		
Max. LG retraction (V _{LOR})	156		
	1700 kg	1785 kg	
Min. Landing speed Flaps UP	85	86	
Min. Landing speed Flaps APP	82	82	
Min. Landing speed Flaps LDG	76 78		
Min. Go-around speed Flaps UP 82 82			
Max. cruising speed (V _{NO})	155		
Never exceed speed (V _{NE})	194		
	up to 1542 kg	above - 1542 kg	
Manoeuvring speed (V _A)	120	126	

MASS		
Max. TKOF mass	1785 kg	
Max ZF mass	1650 kg	
Max. LDG mass	1700 kg	
Empty mass	1295 kg	
Max. baggage in NOSE	30 kg	
Max. baggage in COCKPIT	45 kg	45 kg
Max. baggage in rear EXTENSION	18 kg	45 Kg

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EMERGENCY + ABNORMAL CHECKLIST

For conditions to use this Emergency + Abnormal Checklist see page 1 of the Normal Checklist.

All such conditions are fully applicable also for this checklist.



G1000 Warnings page	2
<u>Engine</u>	
Engine fire / failure during take-off page	4
Engine fire / failure in flight page	4
Engine troubleshooting page	5
Engine restart page	5
Oscillating RPMpage	6
RPM overspeed page	6
<u>Landing Gear</u>	
Landing with defective main gear tire page	6
Landing with defective brakes page	6
Landing gear unsafe warning page	7
Manual extension of landing gear page	7
Landing gear up landing page	7
Smoke and fire	
Engine fire on ground page	
Electrical fire on ground page	8
Electrical fire in flight page	8
Cabin smoke, cabin fire abv 10.000 ft page	9
Other Emergencies	
Oxygen pressure loss above 10.000 ft . page	9
Emergency descent page	9
Suspicion of carbon monoxide page	9
Unintentional flight into icingpage	10
Ice protection failurepage 1	10
<u>Electrical Syste</u>	
Complete electrical failurepage 1	10

G1000 WARNINGS

L/R OIL PRES	Pg. 2	Oil pressure low (red range)
L/R OIL TEMP	Pg. 2	Oil temperature high (red range)
L/R GBOX TEMP	Pg. 2	Gearbox temperature high (red range)
L/R ENG TEMP	Pg. 3	Coolant temperature high (red range)
L/R FUEL TEMP	Pg. 3	Fuel temperature high (red range)
L/R ALTN AMPS	Pg. 3	High Current (red range)
L/R STARTER	Pg. 3	Starter not disengaging
DOOR OPEN	Pg. 3	Unlocked doors
L/R ENG FIRE	Pg. 4 Pg. 4 Pg. 8	Engine fail/fire during take-off Engine fail/fire in flight Engine fire on ground

For other parameters "out of green range" see Abnormal Checklist

Abnormal Checklist starts at page 11

L/R OIL PRES

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OIL PRESSURE LOW

- > Reduce power on affected engine
- > Be prepared for loss of oil and an engine failure; land ASAP

L/R OIL TEMP

OIL TEMPERATURE HIGH

- Check oil pressure
 - **❖** If oil pressure too low (outside green range):
 - ⇒ Reduce power on affected engine
 - ⇒ Expect loss of engine oil
 - ⇒ Be prepared for an engine failure
 - **❖** If oil pressure in green range
 - ⇒ Reduce power on affected engine
 - ⇒ Increase airspeed
 - **❖** If oil temperature not returning to green range:
 - ⇒ Be prepared for an engine failure; land ASAP

L/R GBOX TEMP

GEARBOX TEMPERATURE HIGH

- Reduce power on affected engine
- > Increase airspeed
 - If not returning to green range:
 - ⇒ Be prepared for an engine failure; land ASAP

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Page 2

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L/R ENG TEMP

COOLANT TEMPERATURE HIGH

- > Check G1000 for LOW COOL LVL caution light
 - ❖ If LOW COOL LVL caution light OFF
 - ❖ During climb:
 - ⇒ Reduce power on affected engine by 10% or more as rgrd
 - ⇒ Increase airspeed by 10 KIAS or more as required
 - ⇒ If coolant temp. not returning to green range within 60″: reduce power on affected engine as much as possible and increase airspeed
 - During cruise:
 - ⇒ Reduce power on affected engine
 - ⇒ Increase airspeed
 - ⇒ If coolant temp. not returning to green range: Be prepared for an engine failure; land ASAP
 - If LOW COOL LVL caution light ON
 - ⇒ Reduce power on affected engine
 - ⇒ Expect loss of coolant fluid
 - ⇒ Be prepared for an engine failure

L/R FUEL TEMP

FUEL TEMPERATURE HIGH

- Reduce power on affected engine
- > Increase airspeed
- Transfer fuel from AUX to MAIN tank if applicable
 - If not returning to green range: land ASAP

L/R ALTN AMPS

HIGH CURRENT

- Check circuit breakers
- Reduce electrical load and land ASAP

L/R STARTER

STARTER NOT DISENGAGING

- > Affected power lever IDLE
- > Affected engine master OFF
- > Electric master OFF

DOOR OPEN

UNLOCKED DOORS

- Reduce Airspeed
- Check canopy and rear door visually
 If unable to latch: land ASAP
- Check front baggage doors visually
 - ❖ If one or both open: land ASAP

Never unlatch rear door during flight

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ENGINE FAILURE ENGINE FIRE

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DURING TAKE-OFF

REJECTED TAKE-OFF OR EMERGENCY RE-LANDING

1	Power OFF	1
2	Brakes APPLY	2
3	ATCINFORM	3
	If necessary:	
4	Engine Masters (2) OFF	4
5	Fuel selectors (2) OFF	5
6	Electric Master OFF	6
	In case of fire:	
7	Cabin heat & defrost OFF	7

ENGINE FAILURE ENGINE FIRE

IN FLIGHT

If airspeed below 68 KIAS:

Perform Vmca recovery procedure

Airspeed above 68 KIAS:

	•	
1	Power INCREASE up to MAX	1
2	Airspeedmin Vyse 82 KIAS	2
3	Landing gearUP	3
4	FlapsUP	4
5	Engine Master (dead engine) OFF	5
6	Alternator (dead engine) OFF	6
7	Fuel selector (dead engine) OFF	7
	In case of fire:	
8	Cabin heat & defrost OFF	8
9	Canopy UNLATCH if necessary	9
	Max airspeed 120 KIAS	

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DA42	Twin Star EMERGENCY PROCEI	DURES
	ENGINE TROUBLESHOOTING	
1	Power MAX	1
2	If in icing conditions: Alternate airOPEN	2
4	Fuel quantity	3 4 5 6
8 Circuit breakers		
	ENGINE RESTART	
1 2 3 4 5 6 7	Airspeed	1 2 3 4 5 6 7
8 9 10	If engine started: Power (affected engine) MODERATE Engine instruments check GREEN RANGE Alternator (affected engine)	8 9 10

DATE	TWIII Stai	UNLS
	OSCILLATING RPM	
1	Power lever change setting If no success:	1
2	ECU swap ECU B	2
3	If no success: ECU swapAUTO Land ASAP	3
	RPM OVERSPEED	
1	Power settingREDUCE	1
2	If no success: ECU swapECU B	2
3	If no success: ECU swapAUTO	3
	Land ASAP Be prepared for ENGINE FAILURE IN FLIGHT	
	20 p. op a. ou . o	
LA	NDING WITH DEFECTIVE MAIN GEAR TI	RE
	ATCINFORMED	1
• Lan	For landing:	
	d on RWY side with "good" tire p wing on "good" side low	
	port directional control with brake	
•		
	LANDING WITH DEFECTIVE BRAKES	
1 2 3	After touchdown (if necessary): Engine Masters (2)	1 2 3

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EMERGENCY PROCEDURES

LANDING GEAR UNSAFE WARNING

	If on for more than 20 seconds:	
1	Airspeedmax 156 KIAS	1
	In cold temperature:	
2	Airspeedmax 110 KIAS	2
3	Gear selector RECYCLE	3
	If landing gear extension unsuccessful:	
	Continue with MANUAL EXTENSION	

If landing gear **retraction** unsuccessful:

MANUAL EXTENSION OF LANDING GEAR

Consider flight with landing gear down

1	Airspeedmax 156 KIAS	1
	Gear indicator lightsTEST	2
3	Electric masterCHECK ON	3
4	Bus voltage CHECK NORMAL	4
5	Circuit breaker CHECK	5
6	Gear selector DOWN	6
7	Manual extension handlePULL	7
	If necessary	
8	Airspeedmax 110 KIAS	8
	Apply moderate yawing	
9	Gear indicator lightsCHECK 3 GREENS	9

LANDING GEAR UP LANDING

	(Landing gear completely retracted)	
1	ApproachNORMAL	1
	Just before touchdown:	
2	Power lever IDLE	2
	After touchdown:	
3	Engine Masters (2) OFF	3
4	Fuel selectors (2) OFF	4
5	Electric Master OFF	5

	ENGINE FIRE ON GROUND	
1 2 3 4 5	Power levers (2)	1 2 3 4 5
	ELECTRICAL FIRE ON GROUND	
1 2 3 4 5	Mayday call	1 2 3 4 5
	ELECTRICAL FIRE IN FLIGHT	
1 2 3 4 5 6 7	Emergency switch ON Mayday call CONSIDER Avionic master OFF Electric master OFF Cabin heat & defrost OPEN as necessary Canopy UNLATCH if necessary	1 2 3 4 5 6 7

Max airspeed 120 KIAS

Land ASAP

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DA42	Twin Star EMERGENCY PROCEI	URES
	CABIN SMOKE ABOVE 10.000 FT	
1 2	Oxygen	1 2
3	Oxygen OFF Land ASAP	3
	CABIN FIRE ABOVE 10.000 FT	
1 2	Oxygen	1 2
O	XYGEN PRESSURE LOSS ABOVE 10.000 F	- T
1 2 3	Oxygen	1 2 3
4	Oxygen pressure	
	EMERGENCY DESCENT	
1 2 3 4	Flaps	1 2 3 4
	SUSPICION OF CARBON MONOXIDE	
1 2 3	Cabin heat & defrost OFF Ventilation OPEN Emergency windows OPEN	1 2 3

UNINTENTIONAL FLIGHT INTO ICING Pitot heat ON Cabin heat & defrost ON Power INCREASE Alternate air OPEN as required Emergency windows OPEN as required When pitot heat fails: 7 Alternate static valve OPEN **ICE PROTECTION FAILURE** 1 Airspeed...... MIN 121 KIAS Flaps MAX APP 3 Approach with residual ice 82 KIAS 4 Landing distance x 1,4 **COMPLETE ELECTRICAL FAILURE** 1 Circuit breakers......CHECK all IN If no success: 2 Emergency switch ON Flood light ON 4 Power......SET according power lever position and/or engine noise 5 FlapsVERIFY POSITION Land ASAP Landing gear may slowly extend For landing apply "Manual extension of landing gear"

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EMERGENCY PROCEDURES

G1000 CAUTION LIGHTS

L/R ECU A FAIL	Page 12	ECU A failed
L/R ECU B FAIL	Page 12	ECU B failed
L/R ALTN FAIL	Page 12	Alternator failed
L/R VOLTS LOW	Page 12	Bus voltage too low
L/R COOL LVL	Page 13	Engine coolant level low
PITOT FAIL	Page 13	Pitot heating system failed
PITOT HT OFF	Page 13	Pitot heating system OFF
STALL HT FAIL	Page 13	Stall warning heating failed
STALL HT OFF	Page 13	Stall warning heating OFF
L/R FUEL LOW	Page 13	Main tank fuel qty low
L/R AUX FUEL E	Page 13	L/R auxiliary fuel tank empty
STICK LIMIT	Page 13	Stick limiting system failed
DEICE LVL LO	Page 14	De-icing fluid level low
DEIC PRES LO	Page 14	De-icing pressure low
DEIC PRES HI	Page 14	De-icing pressure high

Engine instrument indications outside of green range

COOLANT temperature high/low	page	15
OIL temperature high/low	page	15
OIL pressure high/low	page	15
FUEL temperature high/low	page	15
VOLT low	page	16
RPM high	nage	16

Other abnormal situations

Both Alternators failed	page	16
Hydraulic pump fail or continuous ops	page	16
AllX fuel transfer fail	nage	16

CAUTION ALERTS ON THE G1000

L/R ECU A OR B FAIL ON GROUND

Discontinue operation, terminate flight preparation

L/R ECU A FAIL

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DURING FLIGHT

Remark: in case of ECU A fail the system automatically switches to ECU B

- Press ECU TEST button for more than 2 seconds
 - If ECU A caution message re-appears or cannot be reset:
 - ⇒ Land ASAP
 - If ECU A caution message can be reset
 - ⇒ Continue flight. Engine must be serviced after LDG

L/R ECU B FAIL

DURING FLIGHT

- Press ECU TEST button for more than 2 seconds
 - If ECU B caution message re-appears or cannot be reset:
 - ⇒ Land ASAP
 - **❖** If ECU B caution message can be reset
 - ⇒ Continue flight. Engine must be serviced after LDG

L/R ALTN FAIL

ALTERNATOR FAILED

- Alternator on affected side OFF
- Monitor bus voltage
- **Reduce electrical consumers**
 - If both alternators failed:
 - ⇒ See Abnormal Checklist "Both Alternators failed", page 16

L/R VOLTS LOW

BUS VOLTAGE TOO LOW

Remark: possible reasons are

- fault in the electrical power supply
- RPM too low
- Continue with "Engine instrument indications outside of green range"
 - VOLTS low, page 16

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L/R COOL LVL

ENGINE COOLANT LEVEL LOW

- Monitor annunciations / engine instruments
- Check coolant temperature
- See "Engine instrument indications outside of green range" -**COOLANT TEMPERATURE**

PITOT FAIL

PITOT HT OFF

STALL HT FAIL

STALL HT OFF

- > check pitot heat ON, if in icing conditions
 - ⇒ expect failure of the pitot-static-system
- \Rightarrow expect loss of aural stall warning
- alternate static valve: OPEN
- leave area with icing conditions (see Emergency Checklist page 10 "Unintentional flight into icing")

L/R FUEL LOW

MAIN TANK FUEL QTY LOW

- Check fuel quantity
 - If LH & RH quantities show remarkable difference:
 - ⇒ Expect loss of fuel on side with lower indication
 - Use x-feed: Fuel selector to x-feed on side with LOW FUEL indication

L/R AUX FUEL E

AUXILIARY FUEL TANK EMPTY

⇒ L/R auxiliary fuel pump OFF

STICK LIMIT

VARIABLE ELEVATOR BACKSTOP SYSTEM FAILED

- ❖ 1 or 2 power levers set for MORE than 20% load:
 - ⇒ Elevator variable backstop is INOP
 - ⇒ Do not stall in any configuration!
- 2 power levers set for LESS than 20% load:
 - ⇒ Elevator variable backstop always ACTIVE
 - ⇒ Reduced elevator capacity
 - ⇒ For approach min VREF 76/78 KIAS

DEICE LVL LO

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DE-ICING FLUIS LEVEL LOW

Maximum duration of ice protection in NORMAL mode: 45 min, in HIGH mode: 22 min

DEIC PRES LO

DE-ICING PRESSURE LOW

- Switch DE-ICE to HIGH
 - If DEIC PRES LO light still ON
 - PUMP1 / PUMP2: select other pump
 - If necessary prime pump by activating windshield pump
 - ❖ If DEIC PRES LO light still ON
 - ⇒ Activate ALTERNATE switch
 - ❖ If DEIC PRES LO light still ON
 - Go to Emergency Checklist page 10 **ICE PROTECTION FAILURE**

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- If DEIC PRES LO light OFF
 - Continue flight (de-icing fluid flow: 30 lt/hr)
 - Monitor ice protection system operation
 - Check de-icing fluid level periodically

DEIC PRES HI

DE-ICING PRESSURE HIGH

- Possible reduced system performance
- Filter cartridge to be replaced at next scheduled maintenance

ENGINE INSTRUMENT INDICATIONS OUTSIDE OF GREEN RANGE

COOLANT temperature high

Refer to Emergency Checklist page 3, "L/R ENG TEMP"

COOLANT temperature low

Remark: During low power descent from high altitude coolant temperature may decrease. Consider increasing power.

- > Check G1000 for LOW COOLANT LVL caution light
- ❖ If "LOW COOLANT LVL caution light" ON
 - ⇒ Reduce power on affected engine
 - ⇒ Expect loss of coolant fluid
 - ⇒ Be prepared for an engine failure

OIL temperature high

Refer to Emergency Checklist page 2, "L/R OIL TEMP"

OIL temperature low

- > Increase power
- Reduce airspeed

OIL pressure high

- Check oil temperature and coolant temperature
 - ❖ If within green range
 - ⇒ Oil pressure indication may be faulty; watch temperatures
 - If outside of green range
 - ⇒ Reduce power on affected engine;
 - ⇒ Be prepared for an engine faiure; Land ASAP

OIL pressure low

Refer to Emergency Checklist page 2, "L/R OIL PRES"

FUEL temperature high

Refer to Emergency Checklist page 3, "L/R FUEL TEMP"

FUEL temperature low

- Increase power on affected engine
- Reduce airspeed
 - If not returning to green range:
 - ⇒ Be prepared for an engine failure; Land ASAP

VOLTS low

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- On ground:
 - ⇒ Check circuit breakers
 - ⇒ Increase RPM
 - ❖ If LOW VOLTS CAUTION still indicated on the G1000:
 - ⇒ Discontinue operation; terminate flight preparation
- In flight:
 - ⇒ Check circuit breakers
 - ⇒ Switch off unnecessary electrical equipment
 - If LOW VOLTS CAUTION still indicated on the G1000:
 - ⇒ Apply L/R ALTN FAIL caution procedure, page 12

RPM high

- Reduce power on affected engine
- Keep RPM in green range with appropriate power lever setting
 - If problem not solved:
 - ⇒ Refer to Emergency Checklist page 6 "RPM overspeed"
 - ⇒ Land ASAP

OTHER ABNORMAL SITUATIONS

Both alternators failed

- Avionic Master: OFF
- > LH/RH Alternator: OFF
- Transponder: STBY
- > Gear: DOWN
 - When down and locked:
 - ⇒ Pull manual gear extension handle
- Stall/Pitot heat: OFF
- All lights:OFF
 - ⇒ Expect battery power to last for 30 minutes
 - ⇒ Expect engine stoppage after this time
 - ⇒ Land ASAP

Hydraulic pump: failure or continuous operation

- Check gear indication lights
- > Prepare for manual landing gear extension

L/R Auxiliary fuel XFER FAIL

- Check fuel quantity
- > Use X-feed to keep LH and RH fuel levels equal
- Amend flight plan to allow for reduced amount of available fuel

FMS Intitialization – AUX 4 page Recommended and compulsory settings

TIME FORMAT	UTC	
NAV ANGLE	AUTO	
DIS. SPD	NAUTICAL	
ALT. VS	FEET	
PRESSURE	HECTOPASCAL	
TEMP	CELSIUS	
FUEL, FF	GALLONS	
POSITION	HDDD°MM.MM′	
MAP DATUM	WGS 84	
AIRSPACE ALERTS	As desired	
ARRIVAL ALERT	As desired	
VOICE	As Desired	

MFD DATA BAR FIELDS	1 GS			
	2 DIS			
	3 ETE			
	4 MSA			
GPS CDI				
SELECTED	AUTO			
ILS CDI CAPTURE	If coupled ILS APCH planned: MANUAL			
COM CHANNEL SPACING	25,0 KHZ			
NEAREST APT				
RWY SURFACE	As desired			
Compulsory:				

PFD Flight Plan	ETA	ESA

MFD Flight Plan	DTK	DIS

ARINC 424 Distance Coding:

Α	В	С	D	Е
1	2	3	4	5
F	G	Н	ı	J
6	7	8	9	10
K	L	M	N	0
11	12	13	14	15
Р	Q	R	S	Т
16	17	18	19	20
U	٧	W	X	Υ
21	22	23	24	25