# TABLE OF CONTENTS

# SECTION 1

# **GENERAL**

Paragra No.	raph	Page No.
1.1	Introduction	1-1
1.3	Engines	
1.5	Propellers	
1.7	Fuel	
1.9		
1.11	Maximum Weights	
1.13	Standard Airplane Weights	
1.15	Baggage Space	
1.17	Specific Loadings	
1.19	Symbols, Abbreviations and Terminology	
1.21	Conversion Factors	1-11

#### SECTION 1

#### **GENERAL**

#### 1.1 INTRODUCTION

This Pilot's Operating Handbook is designed for maximum utilization as an operating guide for the pilot. It includes the material required to be furnished to the pilot by C.A.R. 3 and FAR Part 21 Subpart J. It also contains supplemental data supplied by the airplane manufacturer.

This handbook is not designed as a substitute for adequate and competent flight instruction, knowledge of current airworthiness directives, applicable federal air regulations or advisory circulars. It is not intended to be a guide for basic flight instruction or a training manual and should not be used for operational purposes unless kept in a current status.

Assurance that the airplane is in an airworthy condition is the responsibility of the owner. The pilot in command is responsible for determining that the airplane is safe for flight. The pilot is also responsible for remaining within the operating limitations as outlined by instrument markings, placards, and this handbook.

Although the arrangement of this handbook is intended to increase its in-flight capabilities, it should not be used solely as an occasional operating reference. The pilot should study the entire handbook to familiarize himself with the limitations, performance, procedures and operational handling characteristics of the airplane before flight.

The handbook has been divided into numbered (arabic) sections, each provided with a "finger-tip" tab divider for quick reference. The limitations and emergency procedures have been placed ahead of the normal procedures, performance and other sections to provide easier access to information that may be required in flight. The "Emergency Procedures" Section has been furnished with a red tab divider to present an instant reference to the section. Provisions for expansion of the handbook have been made by the deliberate omission of certain paragraph numbers, figure numbers, item numbers and pages noted as being intentionally left blank.

ISSUED: AUGUST 1, 1975 REPORT: VB-750

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#### WARNING

EXTREME CARE MUST BE EXERCISED TO LIMIT THE USE OF THIS MANUAL TO APPLICABLE AIRCRAFT. THIS MANUAL REVISED AS INDICATED BELOW OR SUBSEQUENTLY REVISED IS VALID FOR USE WITH THE AIRPLANE IDENTIFIED ON THE FACE OF THE TITLE PAGE WHEN OFFICIALLY APPROVED. SUBSEQUENT REVISIONS SUPPLIED BY PIPER AIRCRAFT CORPORATION MUST BE PROPERLY INSERTED.

MODEL PA-32R-300, CHEROKEE LANCE

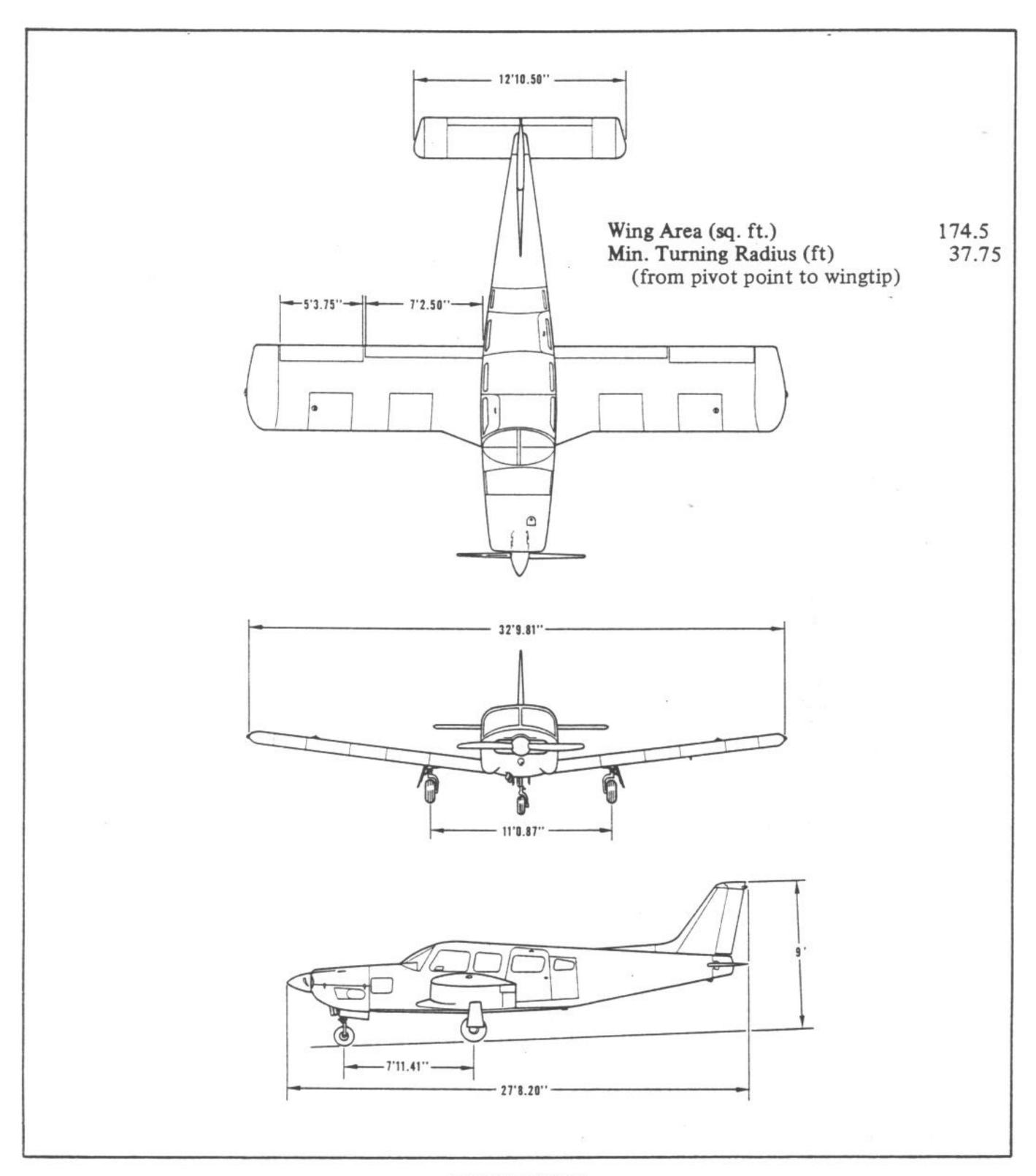
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Barbara J. Morgan

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REPORT: VB-750



THREE VIEW

Figure 1-1

**REPORT**: VB-750 1-2

ISSUED: AUGUST 1, 1975

### 1.3 ENGINES

(a)	Number of Engines		1
(b)	Engine Manufacturer	Lycoming	g
(c)	Engine Model Number	(Serial Nos. 32R-7680001 through 32R-7680140) 10-540-K1A5D	
		(Serial Nos. 32R-7680141 through 32R-7680525) IO-540-K1G5D	
(d)	Rated Horsepower	300	0
(e)	Rated Speed (rpm)	2700	
(f)	Bore (inches)	5.125	
(g)	Stroke (inches)	4.375	
(h)	Displacement (cubic inches)	541.5	
(i)	Compression Ratio	8.7:	
(j)	Engine Type	Six Cylinder, Direct Drive	
		Horizontally Opposed, Air Cooled	

## 1.5 PROPELLERS

(a)	Number of Propellers	1
(b)	Propeller Manufacturer	Hartzell
(c)	Blade Model	F8475D-4
(d)	Number of Blades	2
(e)	Hub Model	HC-C2YK-1( )F
(f)	Propeller Diameter (inches)	
	(1) Maximum	80
	(2) Minimum	78.5
(g)	Propeller Type	Constant Speed,
		Hydraulically Actuated

### 1.7 FUEL

## AVGAS ONLY

(a)	Fuel Capacity (U.S. gal) (total)	98
(b)	Usable Fuel (U.S. gal) (total)	94
(c)	Fuel Grade, Aviation (min. octane)	100/130 - Green

### 1.9 OIL

(a) (b)	Oil Capacity (U.S. quarts) Oil Specification	Refer to	latest issue of
(c)	Oil Viscosity per Average Ambient Temp. for Starting	Lycoming Service In	
	(1) Above 60°F	SINGLE 50	MULTI 40 or 50
	(2) 30°F to 90°F (3) 0°F to 70°F	40	40
	(3) 0 1 10 /0 1	30	40 or 20W-30

ISSUED: AUGUST 1, 1975 REVISED: JULY 13, 1984

(4) Below 10°F

REPORT: VB-750

20

20W-30

1.11 M	AXIMUM WEIGHTS		a
(a) (b)	Maximum Takeoff Weight (lbs) Maximum Landing Weight (lbs)	FORWARD	3600 3600 AFT
(c)	Maximum Weights in Baggage Compartments	100	100
1.13 ST	'ANDARD AIRPLANE WEIGHTS*		
(a)	Standard Empty Weight (lbs): Weight of a standard airplane including unusable fuel,		
(b)	full operating fluids and full oil.  Maximum Useful Load (lbs): The difference		1956
	between the Maximum Takeoff Weight and the Standard Empty Weight.		1644
1.15 BA	AGGAGE SPACE		
(a)	Compartment Volume (cubic feet)	FORWARD 7.0	AFT 20.0
(b) (c)	Entry Width (inches) Entry Height (inches)	16.0 22.0	48.0 26.0
1.17 SP	ECIFIC LOADINGS		
(a) (b)	Wing Loading (lbs per sq ft) Power Loading (lbs per hp)		20.6 12.0

REPORT: VB-750

1-4

ISSUED: AUGUST 1, 1975

**REVISED: FEBRUARY 5, 1979** 

<sup>\*</sup>These values are approximate and vary from one aircraft to another. Refer to Figure 6-5 for the Standard Empty Weight value and the Useful Load value to be used for C.G. calculations for the aircraft specified.

#### 1.19 SYMBOLS, ABBREVIATIONS AND TERMINOLOGY

The following definitions are of symbols, abbreviations and terminology used throughout the handbook and those which may be of added operational significance to the pilot.

# (a) General Airspeed Terminology and Symbols

CAS	Calibrated Airspeed means the indicated speed of an aircraft, corrected for position and instrument error. Calibrated airspeed is equal to true airspeed in standard atmosphere at sea level.
KCAS	Calibrated Airspeed expressed in "Knots."
GS	Ground Speed is the speed of an airplane relative to the ground.
IAS	Indicated Airspeed is the speed of an aircraft as shown on the airspeed indicator when corrected for instrument error. IAS values published in this handbook assume zero instrument error.
KIAS	Indicated Airspeed expressed in "Knots."
M	Mach Number is the ratio of true airspeed to the speed of sound.
TAS	True Airspeed is the airspeed of an airplane relative to undisturbed air which is the CAS corrected for altitude, temperature and compressability.
$V_{\mathbf{A}}$	Maneuvering Speed is the maximum speed at which application of full available aerodynamic control will not overstress the airplane.
$v_{FE}$	Maximum Flap Extended Speed is the highest speed permissible with wing flaps in a prescribed extended position.
$v_{LE}$	Maximum Landing Gear Extended Speed is the maximum speed at which an aircraft can be safely flown with the landing gear extended.

V<sub>LO</sub> Maximum Landing Gear Operating Speed is the maximum speed at which the landing gear can be safely extended or retracted.

V<sub>NE</sub>/M<sub>NE</sub> Never Exceed Speed or Mach Number is the speed limit that may not be exceeded at any time.

Maximum Structural Cruising Speed is the speed that should not be exceeded except in smooth air and then only with caution.

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 $V_{NO}$ 

V<sub>S</sub>
Stalling Speed or the minimum steady flight speed at which the airplane is controllable.

VSO Stalling Speed or the minimum steady flight speed at which the

airplane is controllable in the landing configuration.

V<sub>X</sub>

Best Angle-of-Climb Speed is the airspeed which delivers the greatest gain of altitude in the shortest possible horizontal

distance.

V<sub>Y</sub> Best Rate-of-Climb Speed is the airspeed which delivers the

greatest gain in altitude in the shortest possible time.

# (b) Meteorological Terminology

ISA International Standard Atmosphere in which:

The air is a dry perfect gas;

The temperature at sea level is 15° Celsius (59° Fahrenheit); The pressure at sea level is 29.92 inches hg. (1013.2 mb);

The temperature gradient from sea level to the altitude at which the temperature is -56.5°C (-69.7°F) is -0.00198°C

(-0.003564°F) per foot and zero above that altitude.

OAT
Outside Air Temperature is the free air static temperature,
obtained either from inflight temperature indications or ground
meteorological sources, adjusted for instrument error and

compressibility effects.

Indicated Pressure

The number actually read from an altimeter when the barometric subscale has been set to 29.92 inches of mercury (1013.2 millibars).

Pressure Altitude Altitude measured from standard sea-level pressure (29.92 in. Hg)

altitude corrected for position and instrument error. In this handbook, altimeter instrument errors are assumed to be zero.

by a pressure or barometric altimeter. It is the indicated pressure

Station Pressure Actual atmospheric pressure at field elevation.

Wind The wind velocities recorded as variables on the charts of this

handbook are to be understood as the headwind or tailwind

components of the reported winds.

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#### Power Terminology (Specific) (c)

Takeoff Power

Maximum Rated Power (300 HP @ 2700 RPM)

Maximum Continuous

Power

Maximum Rated Power (300 HP @ 2700 RPM)

Maximum Climb Power

Maximum Rated Power (300 HP @ 2700 RPM)

Maximum Cruise Power

Maximum Rated Power (300 HP @ 2700 RPM)

Flight Idle Power

Throttle Closed

Ground Idle Power

Throttle Closed

(d) Engine Instruments

EGT Gauge

Exhaust Gas Temperature Gauge

Airplane Performance and Flight Planning Terminology

Climb Gradient

The demonstrated ratio of the change in height during a portion of a climb, to the horizontal distance traversed in the same time interval.

Demonstrated Crosswind

Velocity

The demonstrated crosswind velocity is the velocity of the crosswind component for which adequate control of the airplane during takeoff and landing was actually demonstrated during

certification tests.

Accelerate-Stop Distance

The distance required to accelerate an airplane to a specified speed and, assuming failure of an engine at the instant that speed is attained, to bring the airplane to a stop.

Route Segment

A part of a route. Each end of that part is identified by: (1) a geographical location; or (2) a point at which a definite radio fix

can be established.

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REPORT: VB-750

## (f) Weight and Balance Terminology

(C.G.)

Reference Datum An imaginary vertical plane from which all horizontal distances are

measured for balance purposes.

Station A location along the airplane fuselage usually given in terms of

distance from the reference datum.

Arm The horizontal distance from the reference datum to the center of

gravity (C.G.) of an item.

Moment The product of the weight of an item multiplied by its arm.

(Moment divided by a constant is used to simplify balance

calculations by reducing the number of digits.)

Center of Gravity The point at which an airplane would balance if suspended. Its

distance from the reference datum is found by dividing the total

moment by the total weight of the airplane.

C.G. Arm The arm obtained by adding the airplane's individual moments and

dividing the sum by the total weight.

C.G. Limits The extreme center of gravity locations within which the airplane

must be operated at a given weight.

Usable Fuel Fuel available for flight planning.

Unusable Fuel Fuel remaining after a runout test has been completed in

accordance with governmental regulations.

Standard Empty Weight Weight of a standard airplane including unusable fuel, full

operating fluids and full oil.

Basic Empty Weight Standard empty weight plus optional equipment.

Payload Weight of occupants, cargo and baggage.

Useful Load Difference between takeoff weight, or ramp weight if applicable,

and basic empty weight.

Maximum Ramp Weight Maximum weight approved for ground maneuver. (It includes

weight of start, taxi and run up fuel.)

Maximum Takeoff

Weight

Maximum weight approved for the start of the takeoff run.

Maximum Landing

Weight

Maximum weight approved for the landing touchdown.

Maximum Zero Fuel

Weight

Maximum weight exclusive of usable fuel.

REPORT: VB-750

1-8

## 1.21 CONVERSION FACTORS

MULTIPLY	<u>BY</u>	TO OBTAIN	MULTIPLY	<u>BY</u>	TO OBTAIN
acres	0.4047 43560 0.0015625	ha sq. ft. sq. mi.	cubic inches (cu. in.)	16.39 1.639 x 10 <sup>-5</sup> 5.787 x 10 <sup>-4</sup> 0.5541	cm <sup>3</sup> m <sup>3</sup> cu. ft. fl. oz. l U.S. gal. U.S. qt.
atmospheres (atm)	76 29.92 1.0133 1.033	cm Hg in. Hg bar kg/cm <sup>2</sup>		0.01639 4.329 x 10 <sup>-3</sup> 0.01732	
	14.70 2116	lb./sq. in. lb./sq. ft.	cubic meters (m <sup>3</sup> )	61024 1.308 35.3147	cu. in. cu. yd. cu. ft.
bars (bar)	0.98692 14.503768	atm. lb./sq. in.		264.2	U.S. gal.
British Thermal Unit (BTU)	0.2519958	kg-cal	cubic meters per minute (m³/min.)	35.3147	cu. ft./min.
centimeters (cm)	0.3937 0.032808	in. ft.	cubic yards (cu. yd.)	27 0.7646 202	cu. ft. m <sup>3</sup> U.S. gal.
centimeters of mercury at 0°C	0.01316 0.3937	atm in. Hg	degrees (arc)	0.01745	radians
(cm Hg)	0.1934 27.85 135.95	lb./sq. in. lb./sq. ft. kg/m <sup>2</sup>	degrees per second (deg./sec.)	0.01745	radians/sec.
centimeters per	0.032808	ft./sec.	drams, fluid (dr. fl.)	0.125	fl. oz.
second (cm/sec.)	1.9685 0.02237	ft./min. mph	drams, avdp. (dr. avdp.)	0.0625	oz. avdp.
cubic centimeters (cm <sup>3</sup> )	0.03381 0.06102 3.531 x 10 <sup>-5</sup> 0.001 2.642 x 10 <sup>-4</sup>	fl. oz. cu. in. cu. ft. l U.S. gal.	feet (ft.)	30.48 0.3048 12 0.33333 0.0606061 1.894 x 10 <sup>-4</sup>	cm m in. yd. rod
cubic feet (cu.ft.)	28317 0.028317	cm <sup>3</sup>		1.645 x 10 <sup>-4</sup>	mi. NM
	1728 0.037037 7.481 28.32	cu. in: cu. yd. U.S. gal.	feet per minute (ft./min.)	0.01136 0.01829 0.508 0.00508	mph km/hr. cm/sec. m/sec.
cubic feet per minute (cu. ft./min.)	0.472 0.028317	1/sec. m³/min.			

ISSUED: AUGUST 1, 1975

**REVISED: FEBRUARY 5, 1979** 

REPORT: VB-750

1-11

MULTIPLY	<u>BY</u>	TO OBTAIN	MULTIPLY	BY	TO OBTAIN
feet per second (ft./sec.)	0.6818 1.097 30.48 0.5921	mph km/hr. cm/sec. kts.	hectares (ha)	2.471 107639 10000	acres sq. ft. m <sup>2</sup>
foot-pounds (ftlb.)	0.138255 3.24 x 10 <sup>-4</sup>	m-kg kg-cal	horsepower (hp)	33000 550 76.04 1.014	ftlb./min. ftlb./sec. m-kg/sec. metric hp
foot-pounds per minute (ftlb./min.)	3.030 x 10 <sup>-5</sup>	hp	horsepower, metric	75	m-kg/sec.
foot-pounds per second (ftlb./sec.)	1.818 x 10 <sup>-5</sup>	hp	inches (in.)	0.9863 25.40 2.540	mm cm
gallons, Imperial (Imperial gal.)	277.4 1.201 4.546	cu. in. U.S. gal.		0.0254 0.08333 0.027777	m ft. yd.
gallons, U.S. dry (U.S. gal. dry)	268.8 1.556 x 10 <sup>-1</sup> 1.164 4.405	cu. in. cu. ft. U.S. gal.	inches of mercury at 0°C (in. Hg)	0.033421 0.4912 70.73 345.3 2.540	atm lb./sq. in. lb./sq. ft. kg/m² cm Hg
gallons, U.S. liquid (U.S. gal.)	231 0.1337 4.951 x 10 <sup>-3</sup> 3785.4	cu. in. cu. ft. cu. yd.	inch-pounds (inlb.)	25.40 0.011521	mm Hg m-kg
	3.785 x 10 <sup>-3</sup> 3.785 0.83268 128	cm <sup>3</sup> m <sup>3</sup> l Imperial gal. fl. oz.	kilograms (kg)	2.204622 35.27 1000	lb. oz. avdp. g
gallons per acre (gal./acre)	9.353	1/ha	kilogram-calories (kg-cal)	3.9683 3087 426.9	BTU ftlb. m-kg
grams (g)	0.001 0.3527 2.205 x 10 <sup>-3</sup>	kg oz. avdp. lb.	kilograms per cubic meter (kg/m³)	0.06243 0.001	lb./cu. ft. g/cm <sup>3</sup>
grams per centimeter (g/cm)	0.1 6.721 x 10 <sup>-2</sup>	kg/m lb./ft.	kilograms per hectare (kg/ha)	0.892	lb./acre
grams per cubic centimeter (g/cm <sup>3</sup> )	5.601 x 10 <sup>-3</sup> 1000 0.03613 62.43	lb./in.  kg/m³ lb./cu. in. lb./cu. ft.	kilograms per square centimeter (kg/cm <sup>2</sup> )	0.9678 28.96 14.22 2048	atm in. Hg lb./sq. in. lb./sq. ft.

REPORT: VB-750

1-12

ISSUED: AUGUST 1, 1975 REVISED: FEBRUARY 5, 1979